

Dg

EN79-3117,5

FUSELAGE VENTILATION UNDER WIND CONDITIONS

FIRE MODELING AND SCALING METHODS
510-56-05



Jay Wm. Stuart

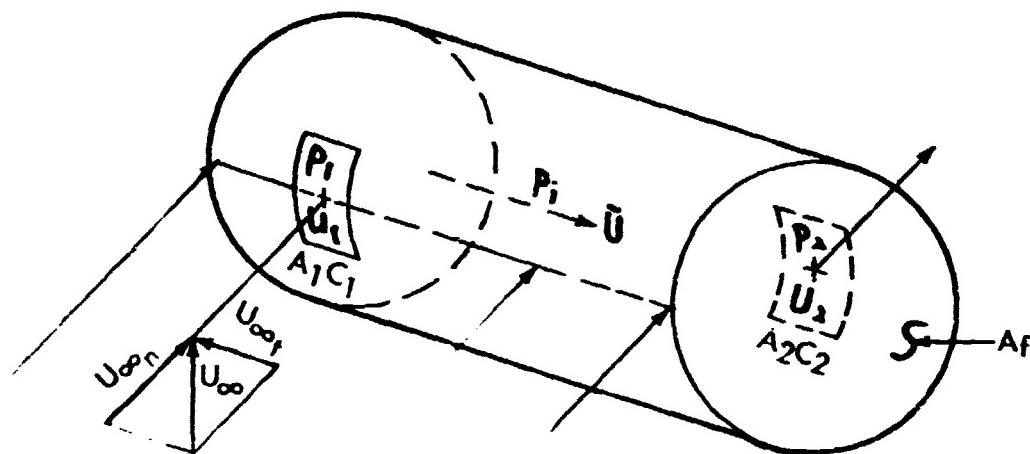


OBJECTIVES

- DETERMINE REALISTIC FUSELAGE VENTILATION RATES FOR POST-CRASH FIRES AND FULL-SCALE FIRE TESTS
- FIND EFFECTS ON WIND-ABOUT-FUSELAGE VENTILATION RATE OF VARIOUS PARAMETERS
 - FUSELAGE SIZE & SHAPE
 - FUSELAGE ORIENTATION & PROXIMITY TO GROUND
 - FUSELAGE-OPENINGS SIZE & LOCATION
 - WIND SPEED & DIRECTION

jpl →

FLUID MECHANICS OF FUSELAGE VENTILATION



FROM MASS CONTINUITY AND ASSUMING $d\rho = 0$

$$\text{SOLVE } U_1 A_1 = U_2 A_2 \quad \text{OR} \quad A_1 C_1 \sqrt{2(p_i - p_f)/\rho} = A_2 C_2 \sqrt{2(p_i - p_f)/\rho}$$

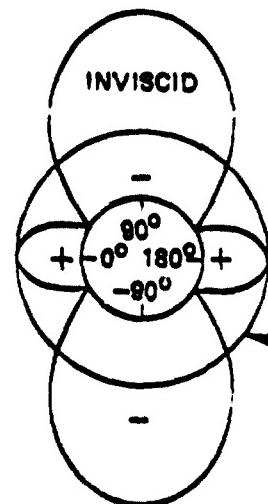
$$\text{LETTING } C_p = p/\rho, \quad q = \frac{\rho}{2} U_{\infty}^2$$

VOLUMETRIC RATE $Q = C_1 A_1 U_{\infty} \sqrt{C_{p1} - \left[C_{p1} + C_{p2} \frac{(A_2 C_2)^2}{(A_1 C_1)} \right] / \left[\frac{(A_2 C_2)^2}{(A_1 C_1)} + 1 \right]}$

INTERIOR VENTILATION SPEED $\bar{U} = Q/A_f$

JPL

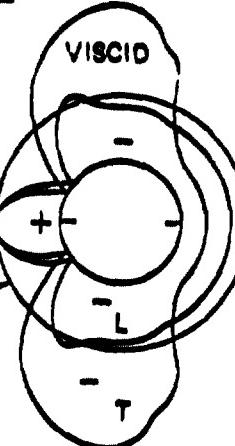
PRESSURE DISTRIBUTIONS FOR FLOWS AROUND INFINITE CIRCULAR CYLINDERS



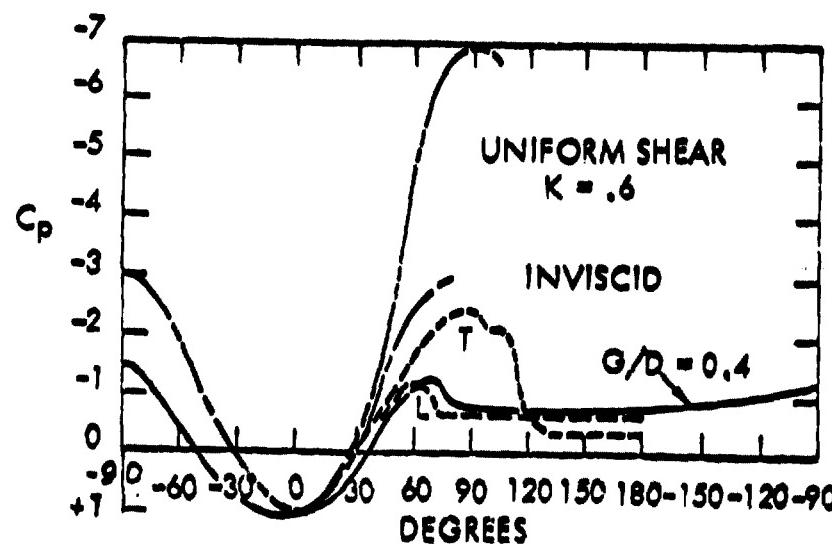
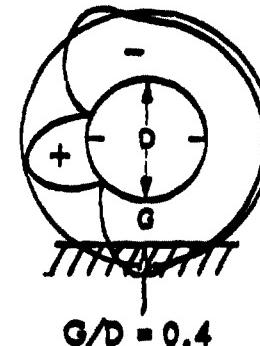
SINGLE CYLINDER

UNIFORM
STREAM

$$C_p = 1.0 \\ \text{TYP}$$



SINGLE CYLINDER IN STREAM PARALLEL TO
WALL WITH BOUNDARY LAYER $R_D = 4.25 \times 10^4$



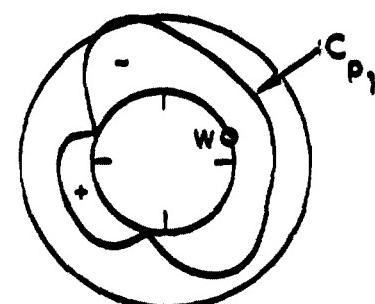
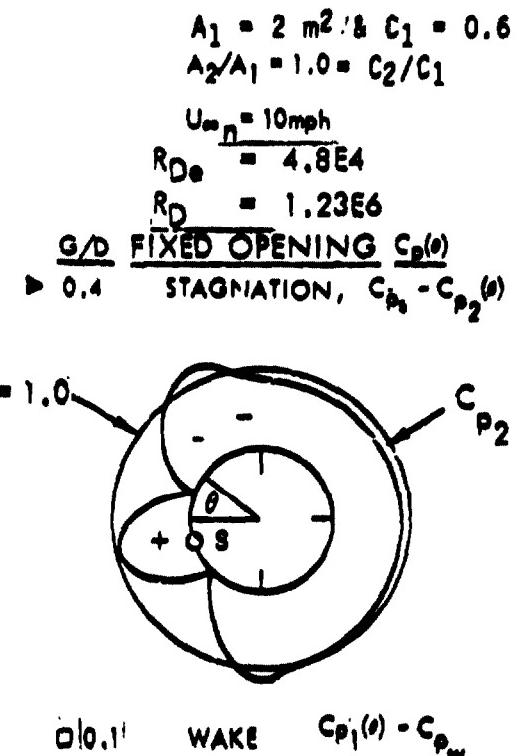
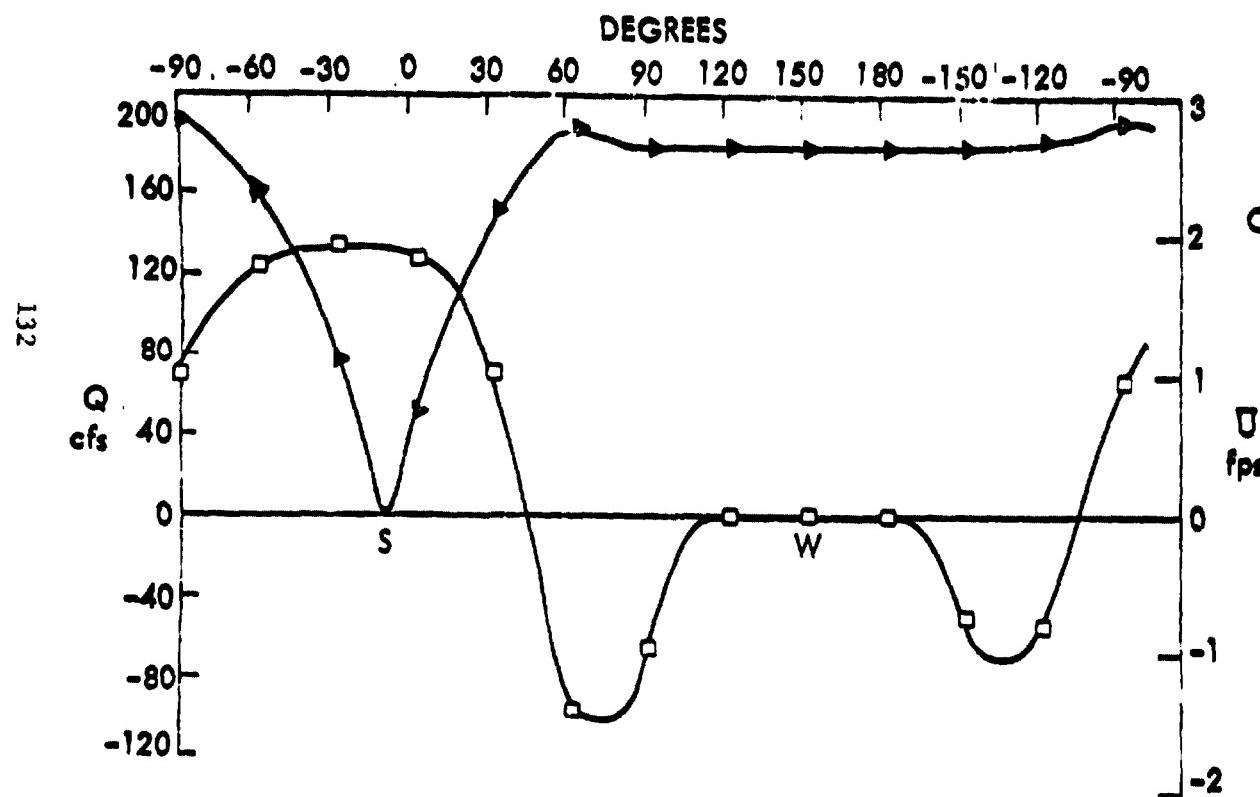


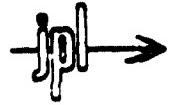
REFERENCES FOR PRESSURE DISTRIBUTIONS AROUND CIRCULAR CYLINDERS

1. FORREST E. GOWEN AND EDWARD W. PERKINS, "DRAG OF CIRCULAR CYLINDERS FOR A WIDE RANGE OF REYNOLDS NUMBERS AND MACH NUMBERS", NACA TN-2960, JUNE 1953
2. MELVIN H. SNYDER JR., "TESTING OF CYLINDERS IN SHEARED FLOW", J. AIRCRAFT, VOL. 8, AUGUST 1971
3. P. W. BEARMAN AND A. J. WADCOCK, "THE INTERACTION BETWEEN A PAIR OF CIRCULAR CYLINDERS NORMAL TO A STREAM", J. FLUID MECH. (1973), VOL. 61, PART 3, PP. 499-511
4. P. W. BEARMAN AND M. M. ZDRAVKOVICH, "FLOW AROUND A CIRCULAR CYLINDER NEAR A PLANE BOUNDARY", J. FLUID MECH. (1978), VOL. 89, PART 1, PP. 33-47

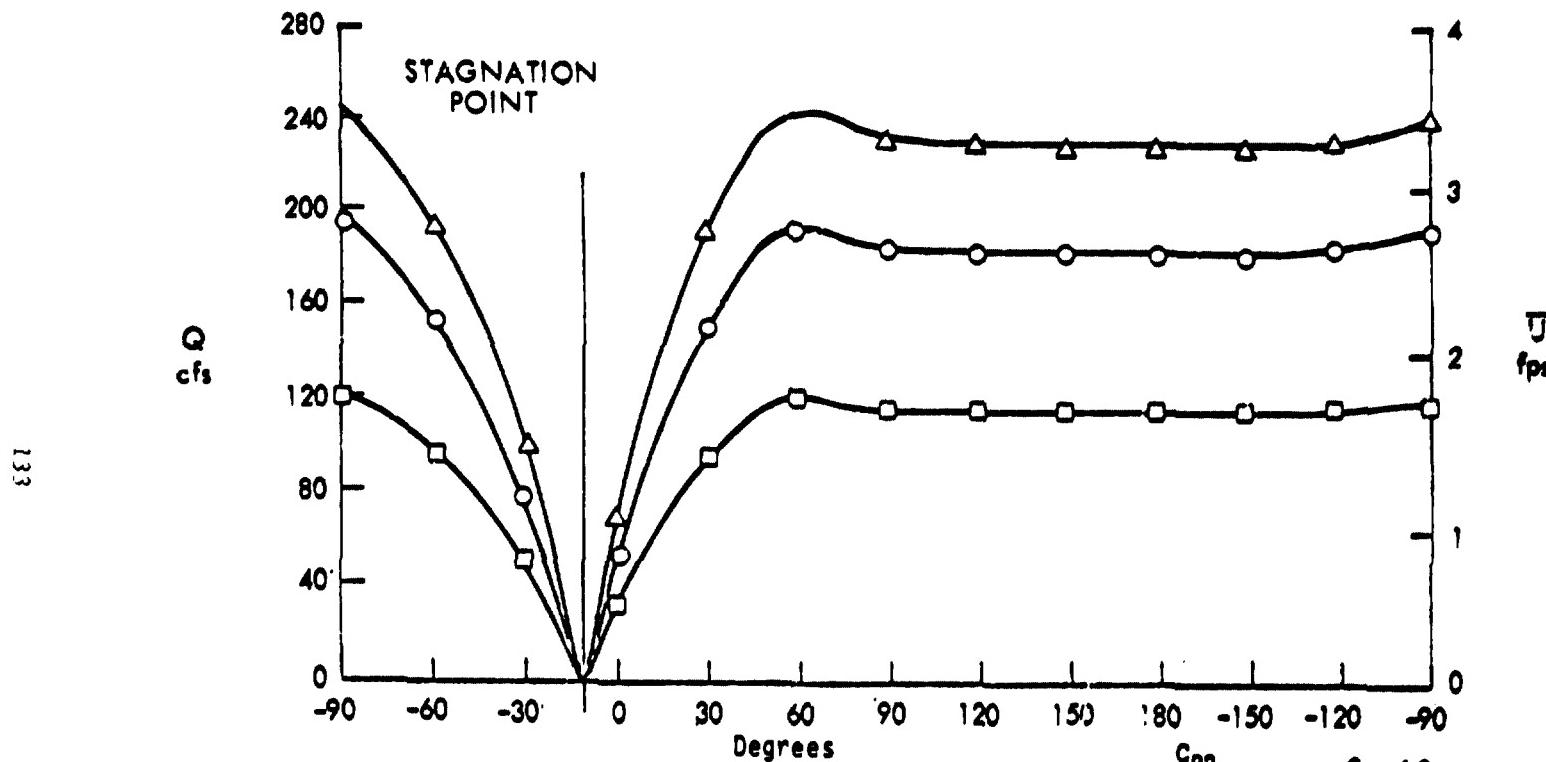


VENTILATION PERFORMANCE COMPARISON FIXED OPENINGS



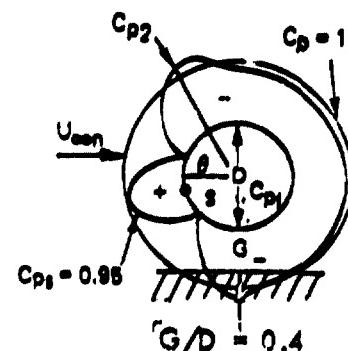


VENTILATION PERFORMANCE IN 2-DIM. FLOW OVER FUSELAGE



$$C_1 = C_2 = 0.6$$

U_∞	= 10 mph	A_2/A_1
A_f	= 2m^2	Δ 2.0
R_{D_0}	= $4.8E4$	\circ 1.0
R_D	= $1.23E6$	\square 0.5





RECOMMENDATIONS

- CONDUCT JSC FULL-SCALE FIRE TESTS TO VALIDATE THE ESTIMATES OF FUSELAGE VENTILATION OF THIS ANALYSIS

- FOR THE REAL WIND-ABOUT-FUSELAGE CONDITIONS EXPERIMENTALLY DETERMINE VENTILATION RATES APPLICABLE TO POST-CRASH FIRES & FULL-SCALE FIRE TESTS

WIND SPEED & DIRECTION

FULL-SCALE REYNOLDS NUMBERS

FUSELAGE SHAPE

FUSELAGE ORIENTATION & PROXIMITY TO GROUND

FUSELAGE-OPENINGS SIZE & LOCATION

FIRE-CONVECTION INDUCED SPEED OR CIRCULATION